

**Warsaw University  
of Technology**



**Faculty of Power and  
Aeronautical Engineering**

WARSAW UNIVERSITY OF TECHNOLOGY

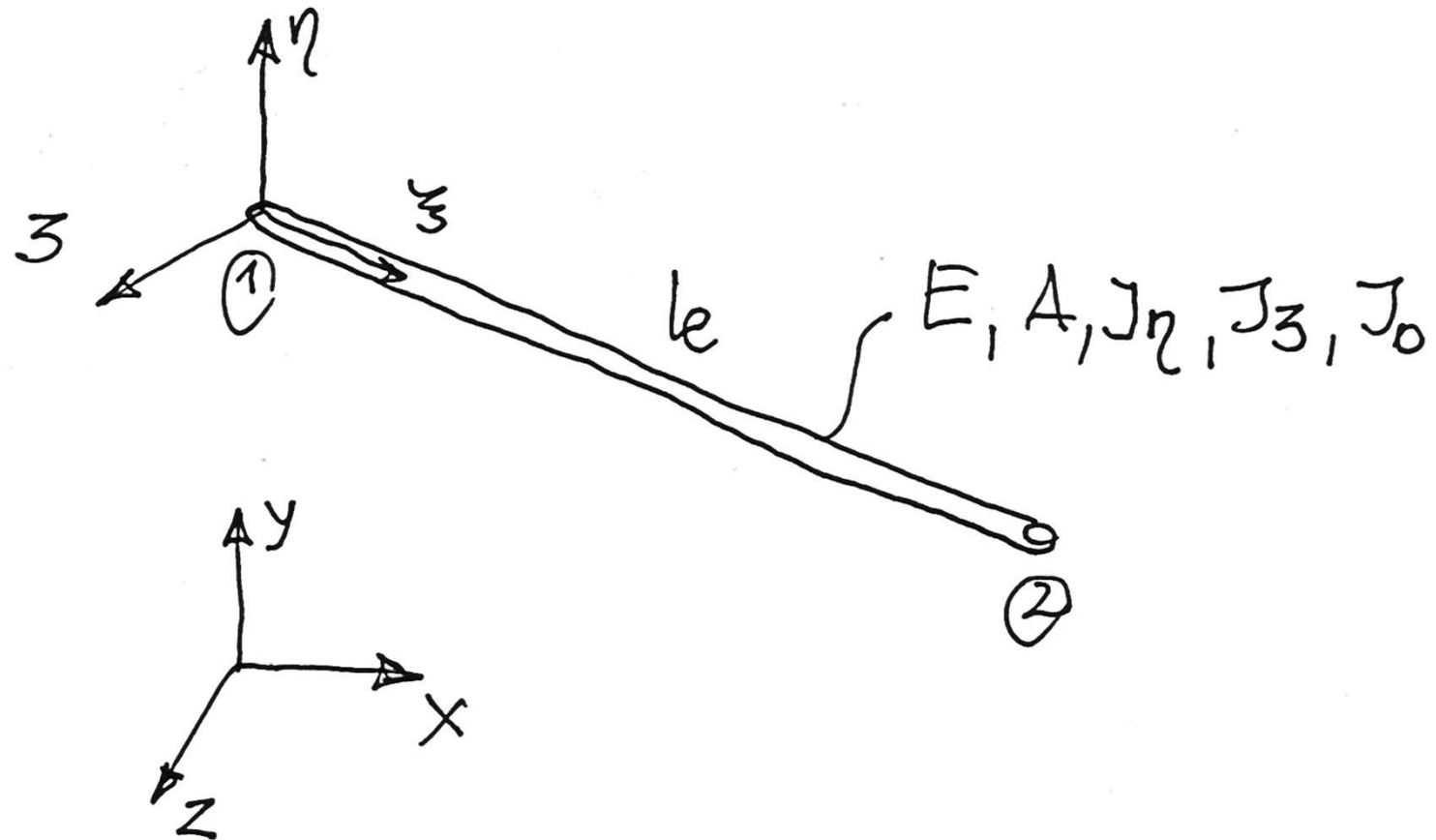
Institute of Aeronautics and Applied Mechanics

# Finite element method (FEM)

3D frame finite element

05.2021

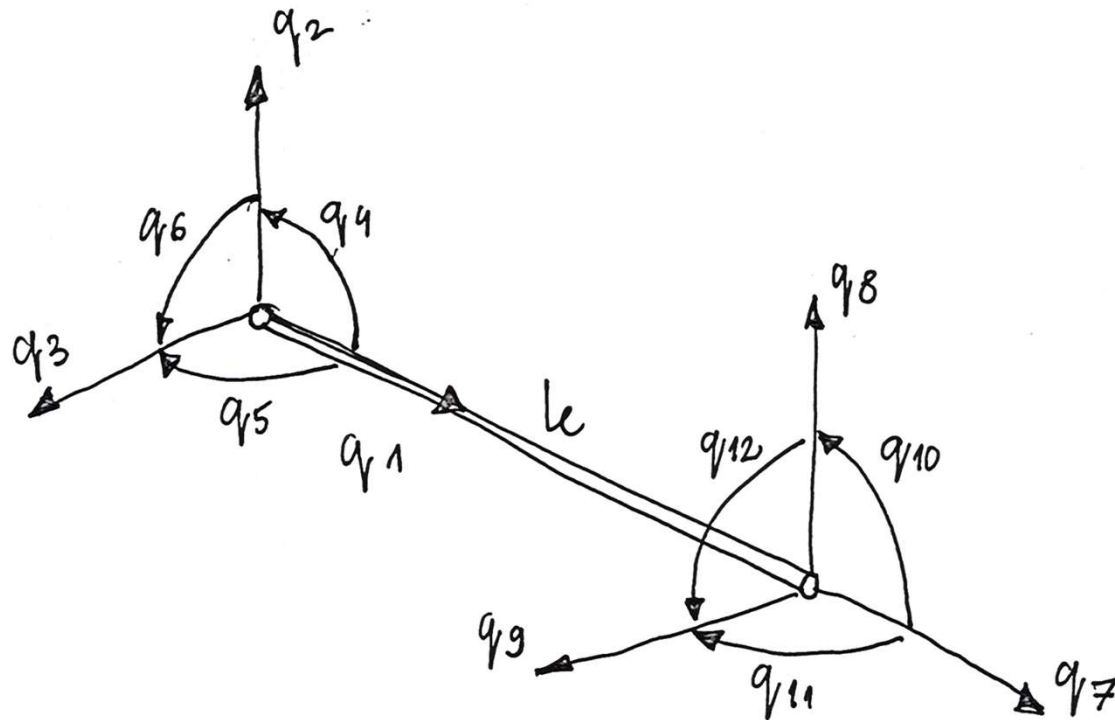
# 3D FRAME ELEMENT



# LOCAL PARAMETERS IN THE COORDINATE SYSTEM §23

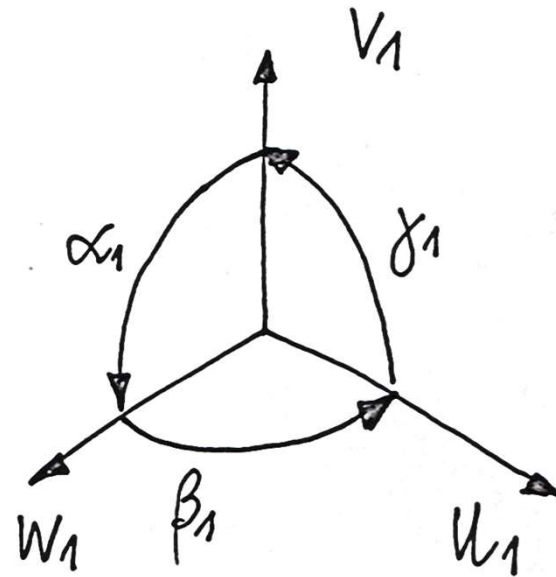
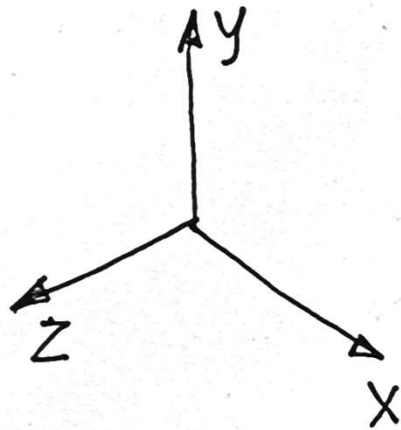
$$L_{q_e} = L_{q_1, q_2, q_3, q_4, q_5, q_6, q_7, q_8, q_9, q_{10}, q_{11}, q_{12}}$$

$1 \times 12$



Local parameters in the coordinate system xyz

$$[g]_{e, 1 \times 12} = [u_1, v_1, w_1, \alpha_1, \beta_1, \gamma_1, u_2, v_2, w_2, \alpha_2, \beta_2, \gamma_2]$$



$$U_e = \frac{1}{2} L^T q_e [k]_e \{q\}_e \quad \text{where :}$$

$1 \times 12$      $12 \times 12$      $12 \times 1$

$$[k]_e =$$

a						-a					
	b		d				-b		d		
		c		e				-c		e	
	d		2r				-d		r		
		e		2s				-e		s	
					t						-t
-a						a					
	-b		-d				b		-d		
		-c		-e				c		-e	
	d		r				-d		2r		
		e		s				-e		2s	
					-t						t

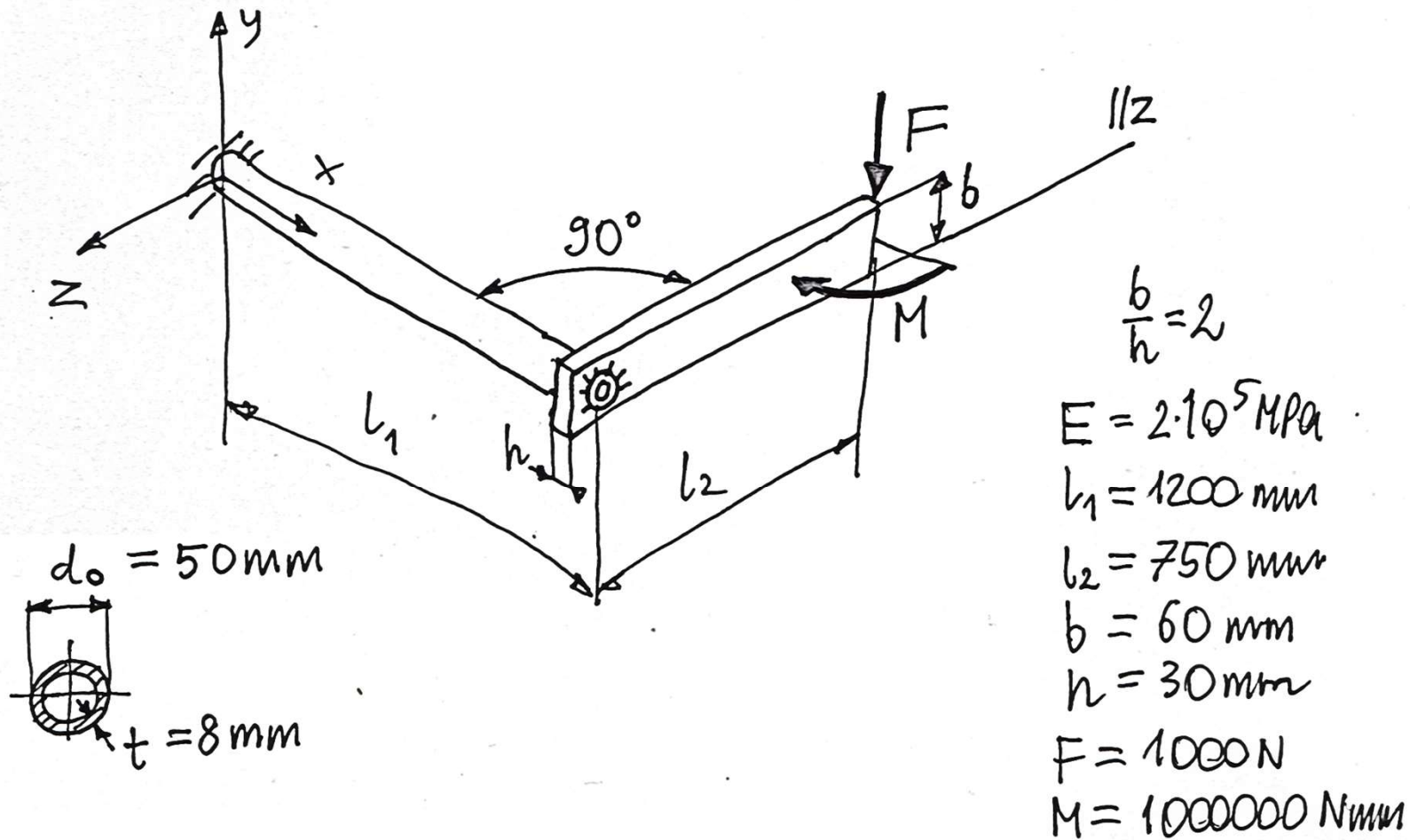
$$a = \frac{EA}{L}, \quad b = \frac{12EJ_3}{l_e^3}, \quad c = \frac{12EJ_2}{l_e^3}, \quad d = \frac{6EJ_3}{l_e^2},$$

$$e = \frac{6EJ_2}{l_e^2}, \quad r = \frac{2EJ_3}{l_e}, \quad s = \frac{2EJ_2}{l_e}, \quad t = \frac{G \cdot J_0}{l_e}$$

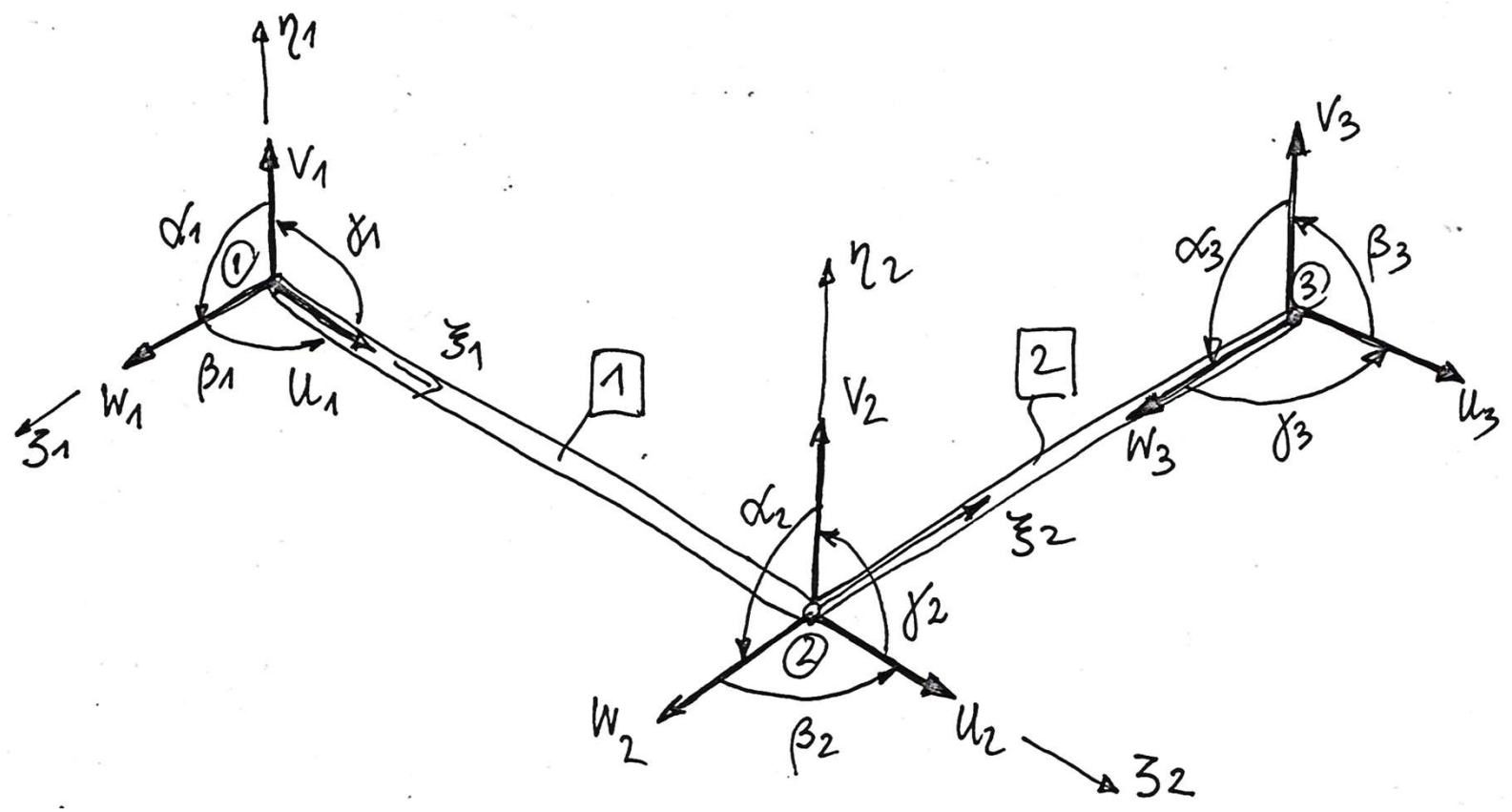
$$U_{ve} = \underbrace{\frac{1}{2} L \underbrace{q_g}_{1 \times 12} \cdot \underbrace{[T_f]^T}_{12 \times 12} \cdot \underbrace{[k]_e}_{12 \times 12} \cdot \underbrace{[T_f]}_{12 \times 12} \cdot \underbrace{\{q_g\}_e}_{12 \times 1}}_{[k_g]_e}$$

$$12 \times 12$$

EXAMPLE : BUILD A FE MODEL USING 3D FRAME ELEMENTS . FIND UNKNOWN DISPLACEMENTS, STRESSES AND REACTIONS .



# Global parameters in the coordinate system xyz

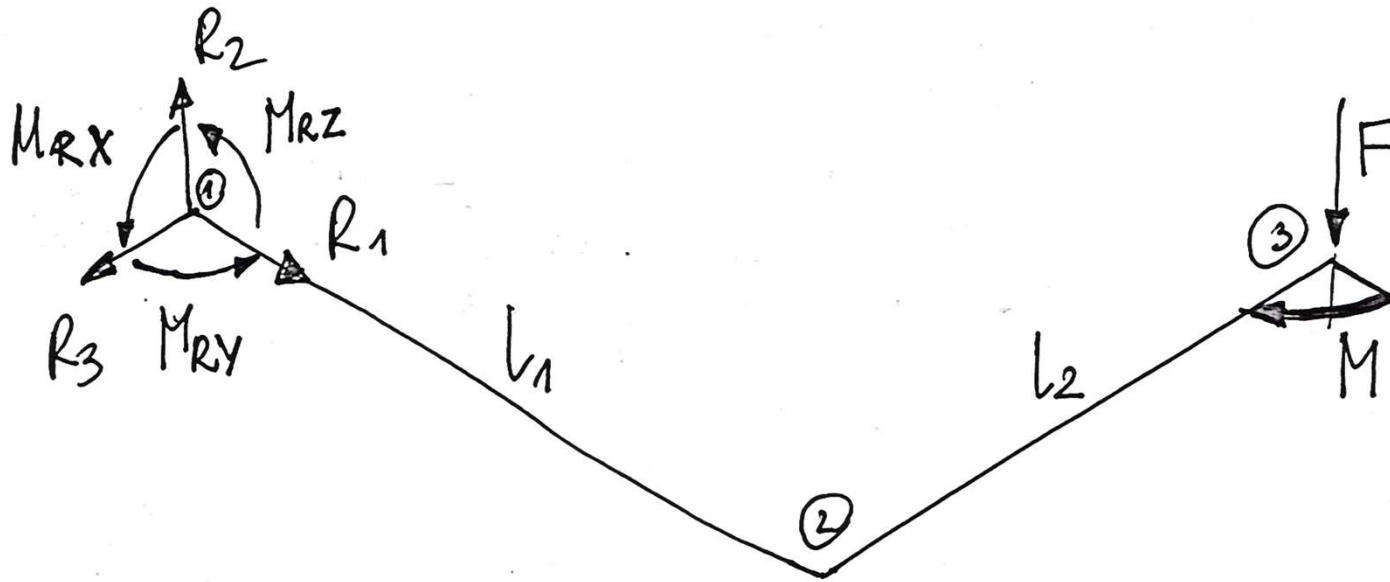


$$Lq = [u_1, v_1, w_1, \alpha_1, \beta_1, \gamma_1, u_2, v_2, w_2, \alpha_2, \beta_2, \gamma_2, u_3, v_3, w_3, \alpha_3, \beta_3, \gamma_3]$$

1x18



# Global load vector



$$[F] = [R_1, R_2, R_3, M_{RX}, M_{RY}, M_{RZ}, 0, 0, 0, 0, 0, 0, -F, 0, -M, 0]$$

$1 \times 16$

ELEMENT 1:

$$[q]_1 = [q_1, q_2, q_3, q_4, q_5, q_6, q_7, q_8, q_9, q_{10}, q_{11}, q_{12}]$$

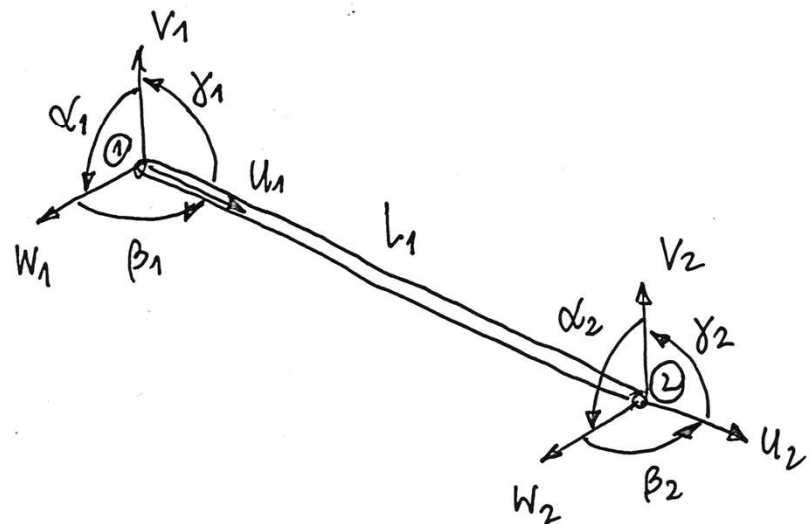
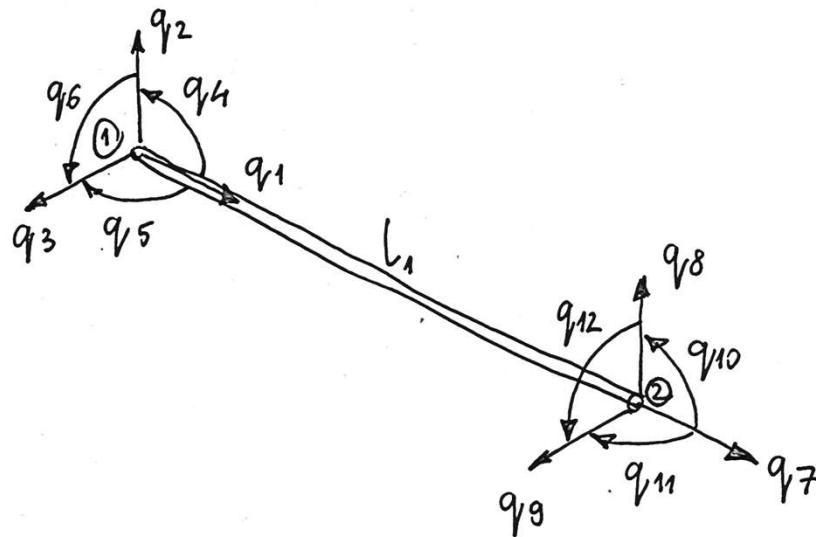
1x12

$$[q_g]_1 = [u_1, v_1, w_1, \alpha_1, \beta_1, \gamma_1, u_2, v_2, w_2, \alpha_2, \beta_2, \gamma_2]$$

1x12

$$q_1 = u_1, \quad q_2 = v_1, \quad q_3 = w_1, \quad q_4 = \gamma_1, \quad q_5 = -\beta_1, \quad q_6 = \alpha_1$$

$$q_7 = u_2, \quad q_8 = v_2, \quad q_9 = w_2, \quad q_{10} = \gamma_2, \quad q_{11} = -\beta_2, \quad q_{12} = \alpha_2$$





$$[k_g]_1 = [\bar{T}_f]_1^T \cdot [k]_1 \cdot [T_f]_1$$

$12 \times 12$        $12 \times 12$        $12 \times 12$        $12 \times 12$

$$d_1 = \frac{EA_1}{L_1}, \quad b_1 = \frac{12EJ_{31}}{L_1^3}, \quad c_1 = b_1, \quad d_1 = \frac{6EJ_{31}}{L_1^2}, \quad e_1 = d_1,$$

$$r_1 = \frac{2EJ_{31}}{L_1}, \quad s_1 = r_1, \quad t_1 = \frac{GJ_{01}}{L_1}, \quad A_1 = \frac{\pi(d_0^2 - (d_0 - 2t)^2)}{4}$$

$$J_{31} = J_{21} = \frac{\pi}{64} (d_0^4 - (d_0 - 2t)^4), \quad J_{01} = 2 \cdot J_{31}$$

$$[k_g]_1^* = \begin{bmatrix} [k_g]_1 & [0] \\ [0] & [0] \end{bmatrix}$$

$18 \times 18$        $12 \times 6$        $6 \times 12$        $6 \times 6$

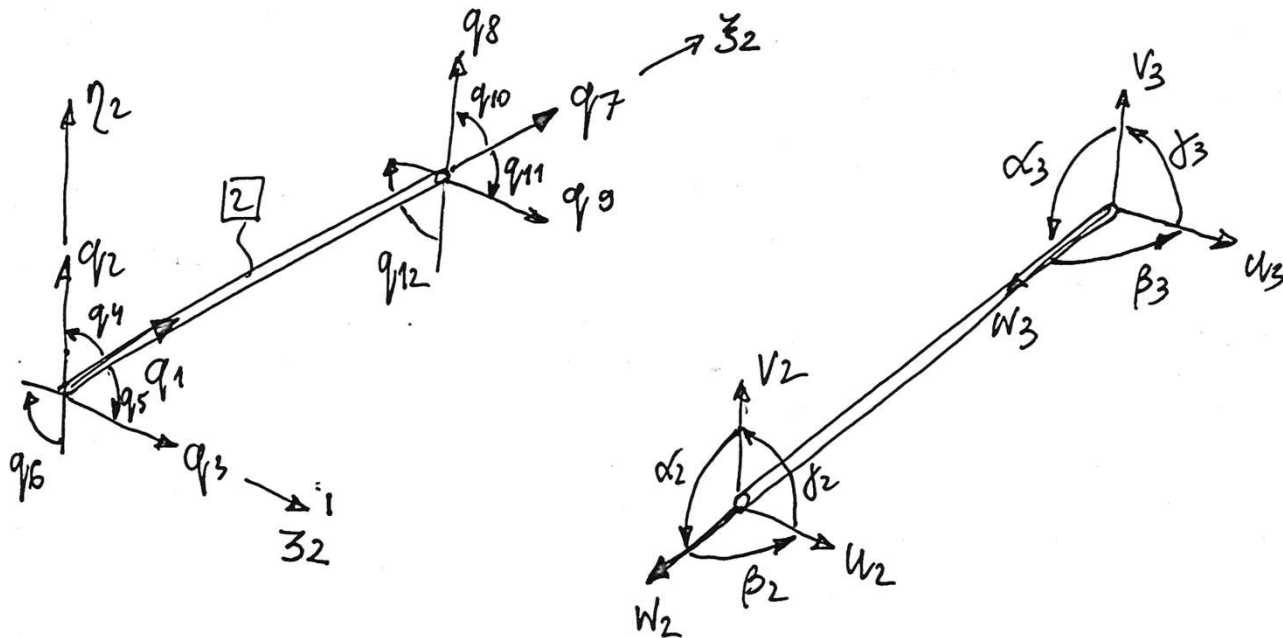
ELEMENT 2

$$[q]_2 = [q_1, q_2, q_3, q_4, q_5, q_6, q_7, q_8, q_9, q_{10}, q_{11}, q_{12}]$$

$$[q]_2 = [u_2, v_2, w_2, \alpha_2, \beta_2, \gamma_2, u_3, v_3, w_3, \alpha_3, \beta_3, \gamma_3]$$

$$q_1 = -w_2, \quad q_2 = v_2, \quad q_3 = u_2, \quad q_4 = \alpha_2, \quad q_5 = -\beta_2, \quad q_6 = -\gamma_2$$

$$q_7 = -w_3, \quad q_8 = v_3, \quad q_9 = u_3, \quad q_{10} = \alpha_3, \quad q_{11} = -\beta_3, \quad q_{12} = -\gamma_3$$



$$\begin{matrix} \{q\}_2 & = & [T_f]_2 \cdot \{q_9\}_2 \\ 12 \times 1 & & 12 \times 12 \quad 12 \times 1 \end{matrix}$$

$$[T_f]_2 = \begin{bmatrix} 0 & 0 & -1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & -1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & -1 & 0 & 0 & 0 & 0 & 0 & 0 \\ \hline & & & & & & 0 & 0 & -1 & 0 & 0 & 0 \\ & & & & & & 1 & 0 & 0 & 0 & 0 & 0 \\ & & & & & & 1 & 0 & 0 & 0 & 0 & 0 \\ & & & & & & 1 & 0 & 0 & 0 & 1 & 0 & 0 \\ & & & & & & 1 & 0 & 0 & 0 & 0 & -1 & 0 \\ \hline & & & & & & 0 & 0 & 0 & 0 & 0 & 0 & -1 \end{bmatrix}$$

$\begin{bmatrix} 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \end{bmatrix}$   
 $6 \times 6$

$$[k_g]_2 = [T_f]_2^T \cdot [K]_2 \cdot [T_f]_2$$

$12 \times 12$        $12 \times 12$        $12 \times 12$        $12 \times 12$

$$a_2 = \frac{EA_2}{l_2}, \quad b_2 = \frac{12EJ_{32}}{l_2^3}, \quad c_2 = \frac{12EJ_{\eta 2}}{l_2^3}, \quad d_2 = \frac{6EJ_{32}}{l_2^2}$$

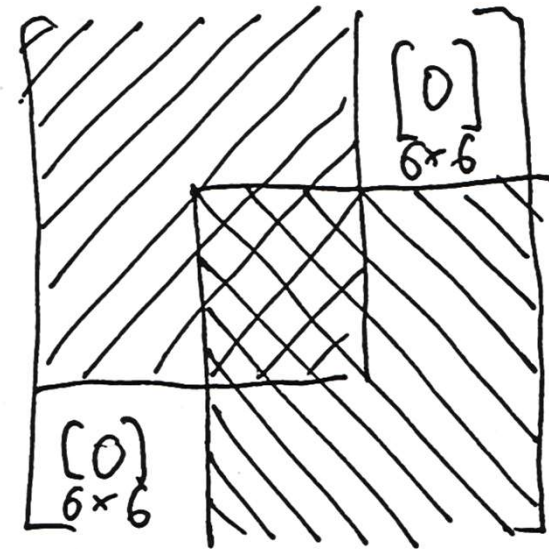
$$e_2 = \frac{6EJ_{\eta 2}}{l_2^2}, \quad r_2 = \frac{2EJ_{32}}{l_2}, \quad s_2 = \frac{2EJ_{\eta 2}}{l_2}, \quad t_2 = \frac{G \cdot J_{o2}}{l_2}$$

$$J_{32} = \frac{hb^3}{12}, \quad J_{\eta 2} = \frac{bh^3}{12}, \quad J_{o2} = 0.457 bh^3$$

$$A_2 = b \cdot h$$

$$[k_g]_2^* = \begin{bmatrix} [0]_{6 \times 6} & [0]_{6 \times 12} \\ [0]_{12 \times 6} & [k_g]_2 \end{bmatrix}$$

$$\begin{matrix} [K] \\ 18 \times 18 \end{matrix} = \begin{matrix} [Kg]_1^* \\ 18 \times 18 \end{matrix} + \begin{matrix} [Kg]_2^* \\ 18 \times 18 \end{matrix} =$$

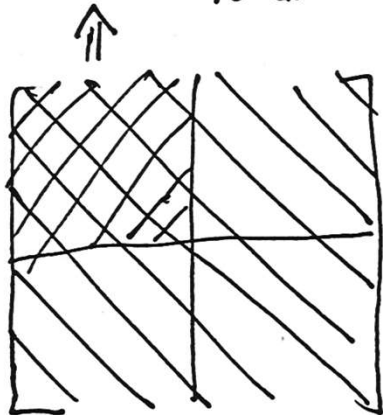




$$\begin{matrix} [K] \cdot \{q\} = \{F\} \\ 18 \times 18 & 18 \times 1 & 18 \times 1 \end{matrix}$$

$$\begin{matrix} u_1 = 0, u_4 = 0, w_1 = 0 \\ v_1 = 0, \beta_1 = 0, \gamma_1 = 0 \end{matrix}$$

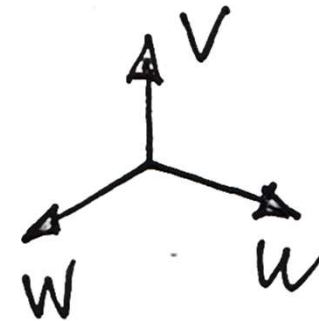
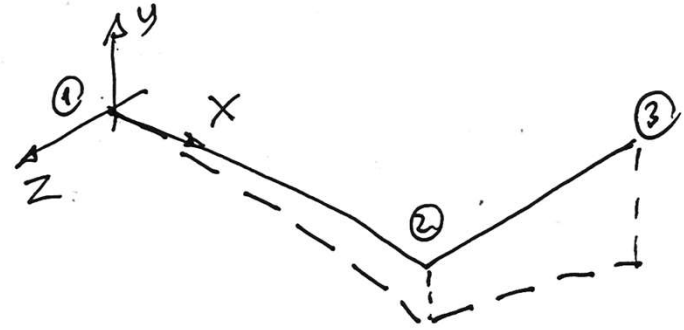
$$\begin{matrix} [K] \cdot \{q\} = \{F\} \\ 12 \times 12 & 12 \times 1 & 12 \times 1 \end{matrix} \Rightarrow \begin{matrix} \{q\} = [K]^{-1} \cdot \{F\} \\ 12 \times 1 & 12 \times 12 & 12 \times 1 \end{matrix}$$



$$\begin{matrix} [K] \cdot \{q\} = \{F\} \\ 18 \times 18 & 18 \times 1 & 18 \times 1 \end{matrix} \Rightarrow \text{REACTIONS}$$

## DOF SOLUTION

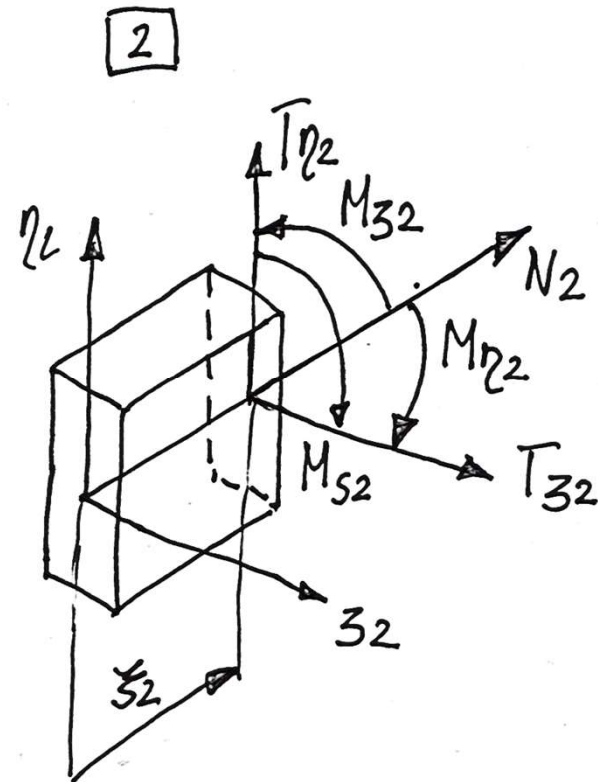
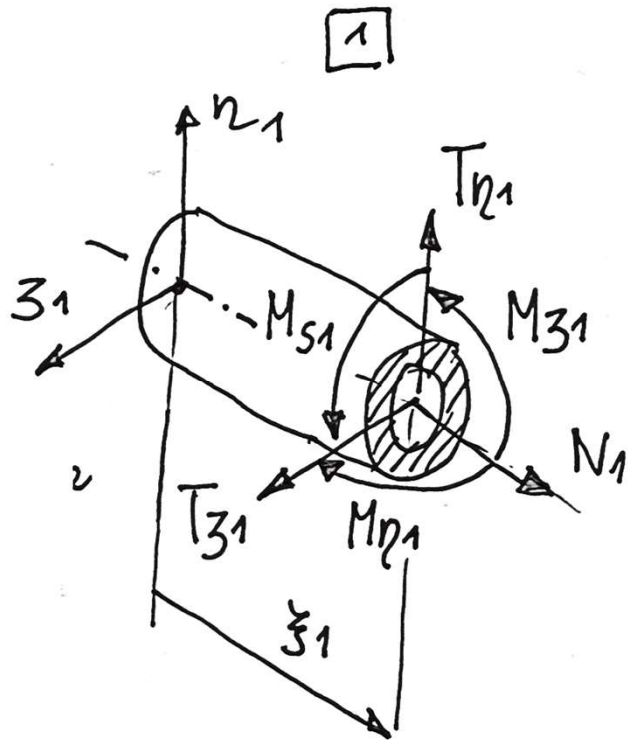
$$\begin{matrix} \{q\} \\ 12 \times 1 \end{matrix} = \begin{Bmatrix} u_2 \\ v_2 \\ w_2 \\ \alpha_2 \\ \beta_2 \\ \gamma_2 \\ u_3 \\ v_3 \\ w_3 \\ \alpha_3 \\ \beta_3 \\ \gamma_3 \end{Bmatrix} = \begin{Bmatrix} 0 \text{ mm} \\ -11.94 \text{ mm} \\ 14.93 \text{ mm} \\ -0.0243 \text{ rad} \\ -0.0249 \text{ rad} \\ -0.015 \text{ rad} \\ 29.07 \text{ mm} \\ -31.43 \text{ mm} \\ 14.93 \text{ mm} \\ -0.0269 \text{ rad} \\ -0.0527 \text{ rad} \\ -0.015 \text{ rad} \end{Bmatrix}$$





# ELEMENT SOLUTION

$$\{q_i\}_{12 \times 1} = [T_i]_{12 \times 12} \cdot \{q_j\}_{12 \times 1}, \quad i = 1, 2$$



AXIAL BAR:  $(q_1, q_7)_i$

$$\varepsilon_{\xi_i} = \frac{(q_7 - q_1)_i}{l_i}, \quad \sigma_{\xi_i} = E \cdot \varepsilon_{\xi_i}, \quad N_i = \sigma_{\xi_i} A_i$$

BEAM:

I) BENDING IN  $(\xi \eta)_i$  PLANE:  $(q_2, q_4, q_8, q_{10})_i$

$$v_i(\xi_i) = \underset{1 \times 4}{[N(\xi_i)]} \cdot \begin{Bmatrix} q_2 \\ q_4 \\ q_8 \\ q_{10} \end{Bmatrix}_i$$

$$\begin{aligned} M_{\xi_i}(\xi_i) &= E \cdot J_{\xi_i} \cdot v_i''(\xi_i) = \\ &= E \cdot J_{\xi_i} \cdot \underset{1 \times 4}{[N''(\xi_i)]} \cdot \begin{Bmatrix} q_2 \\ q_4 \\ q_8 \\ q_{10} \end{Bmatrix}_i \end{aligned}$$

$$\sigma_{\xi_i}^I = - \frac{M_{3i}(\xi_i) \cdot \eta_i}{J_{3i}}$$

$$T_{\eta_i}(\xi_i) = - EJ_{3i} \cdot v_i'''(\xi_i) =$$

$$= - EJ_{3i} \cdot \underset{1 \times 4}{[N''']} \cdot \begin{Bmatrix} q_2 \\ q_4 \\ q_8 \\ q_{10} \end{Bmatrix}_i = \text{const}$$

SHEAR STRESS CAUSED BY  $T_{\eta_i}(\xi_i)$  IS NEGLECTED.

II) BENDING IN ( $\xi_3$ ) PLANE :  $(q_3, q_5, q_9, q_{11})_i$

$$w_i(\xi_i) = \underbrace{[N(\xi_i)]}_{1 \times 4} \cdot \begin{Bmatrix} q_3 \\ q_5 \\ q_9 \\ q_{11} \end{Bmatrix}_i$$

$$M_{\eta_i}(\xi_i) = E \cdot J_{\eta_i} w_i''(\xi_i) =$$

$$= E J_{\eta_i} \cdot \underbrace{[N''(\xi_i)]}_{1 \times 4} \cdot \begin{Bmatrix} q_3 \\ q_5 \\ q_9 \\ q_{11} \end{Bmatrix}_i$$

$$\sigma_{\xi_i}^{\text{II}} = - \frac{M_{\eta_i}(\xi_i) \cdot z_i}{J_{\eta_i}}$$

$$T_{z_i}(\xi_i) = - E J_{\eta_i} \cdot w_i'''(\xi_i) =$$

$$= - E J_{\eta_i} \left[ N_{1 \times 4}''' \right] \cdot \begin{Bmatrix} q_3 \\ q_5 \\ q_9 \\ q_{11} \end{Bmatrix}_i = \text{const}$$

SHEAR STRESS CAUSED BY  $T_{z_i}(\xi_i)$  IS NEGLECTED.



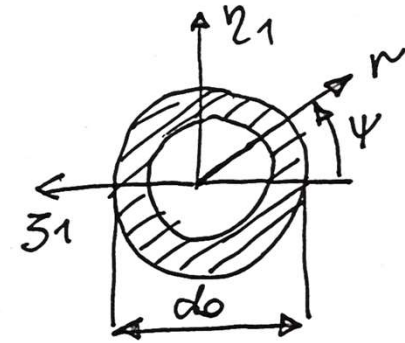
TORSION BAR :  $(q_6, q_{12})_i$

$$\begin{aligned}\varphi_i(\xi_i) &= \underset{1 \times 2}{[N(\xi_i)]} \cdot \begin{Bmatrix} q_6 \\ q_{12} \end{Bmatrix}_i = \left[ 1 - \frac{\xi_i}{l_i}, \frac{\xi_i}{l_i} \right] \cdot \begin{Bmatrix} q_6 \\ q_{12} \end{Bmatrix}_i = \\ &= (q_6)_i + \frac{(q_{12} - q_6)_i}{l_i} \cdot \xi_i\end{aligned}$$

$$\begin{Bmatrix} q_6 \\ q_{12} \end{Bmatrix}_2 = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix}$$

ELEMENT [1] :

$$\begin{aligned}\tau_1(r) &= G \cdot \gamma_1(r) = \frac{E}{2(1+\nu)} \cdot \frac{d\varphi_1(\xi_1)}{d\xi_1} \cdot r = \\ &= \frac{E}{2(1+\nu)} \cdot \left[ -\frac{1}{l_1}, \frac{1}{l_1} \right] \cdot \begin{Bmatrix} q_6 \\ q_{12} \end{Bmatrix}_1 \cdot r = \\ &= \frac{E(q_{12} - q_6)_1}{2(1+\nu)l_1} \cdot r \quad , (q_6)_1 = 0\end{aligned}$$



$$\tau_{1\max} = \tau_1\left(\frac{d_0}{2}\right) = \frac{E \cdot (q_{12})_1 \cdot d_0}{4(1+\nu)l_1} = \frac{E d_0 \alpha_2}{4(1+\nu)l_1}$$

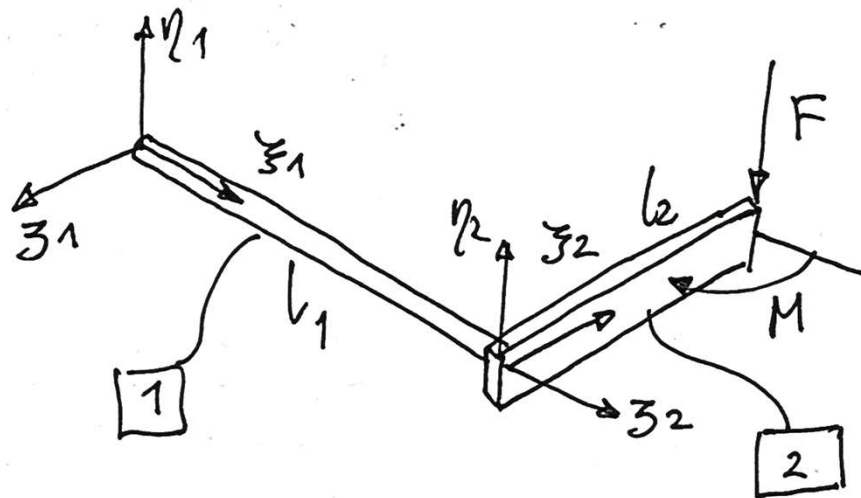
$$M_{S1} = \frac{\tau_1(r) \cdot J_{O1}}{r} = \frac{E \cdot (q_{12})_1 \cdot J_{O1}}{2(1+\nu)l_1} = \frac{E \alpha_2 J_{O1}}{2(1+\nu)l_1} = \text{const}$$

ELEMENT  $\boxed{2}$  :

$$\begin{Bmatrix} q_{16} \\ q_{12} \end{Bmatrix}_2 = \begin{Bmatrix} \delta_2 \\ \delta_3 \end{Bmatrix} = \begin{Bmatrix} -0.015 \text{ rad} \\ -0.015 \text{ rad} \end{Bmatrix} \Rightarrow \varphi_2(\xi_2) = q_{16} = \text{const}$$

$$\Rightarrow \frac{d\varphi_2(\xi_2)}{d\xi_2} = 0 \Rightarrow \begin{aligned} \tilde{L}_2 &= 0 \\ M_{S2} &= 0 \end{aligned}$$

## ELEMENT RESULTS



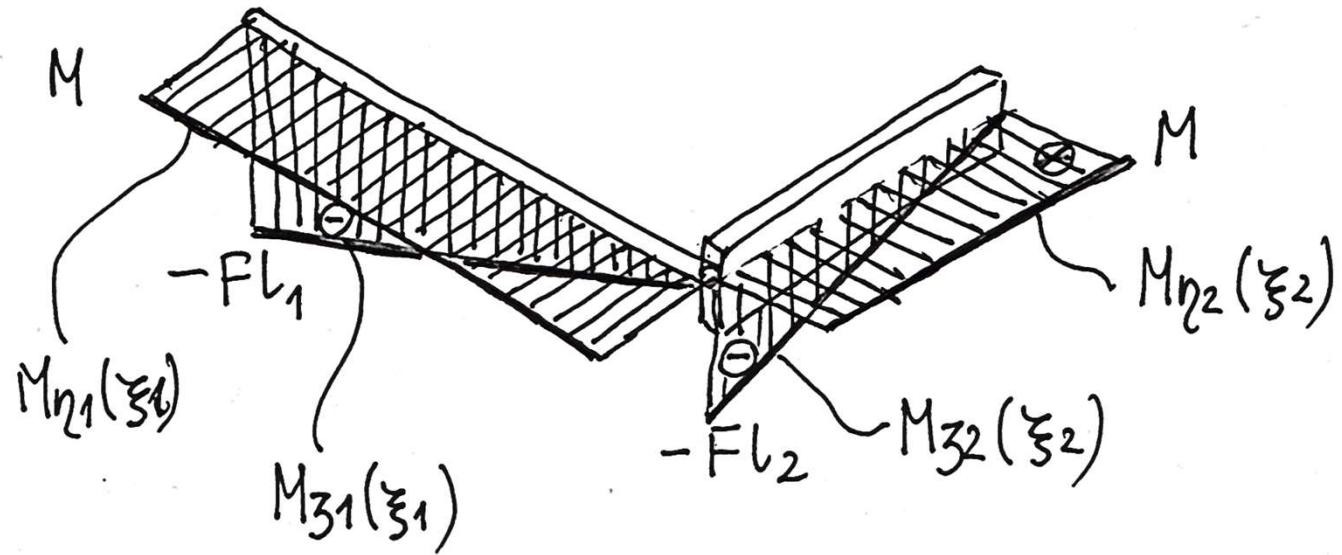
NORMAL FORCES :

$$N_1 = 0 \quad , \quad N_2 = 0$$

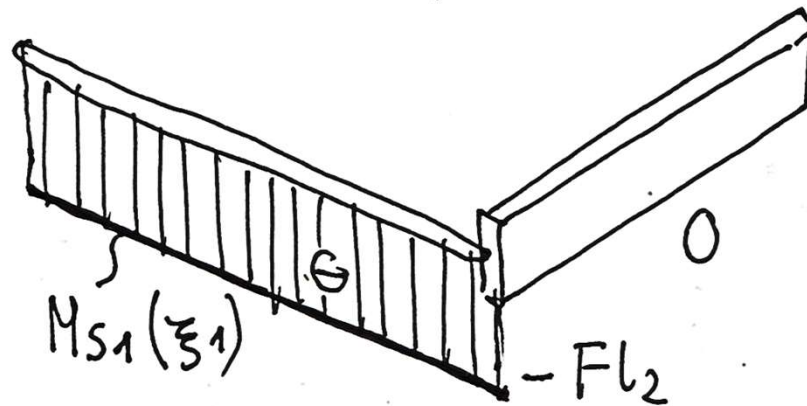
SHEAR FORCES :

$$\begin{aligned} T_{\eta_1} &= -F \quad , \quad T_{\eta_2} = -F \\ T_{z_1} &= 0 \quad , \quad T_{z_2} = 0 \end{aligned}$$

BENDING MOMENTS :



TORQUE :



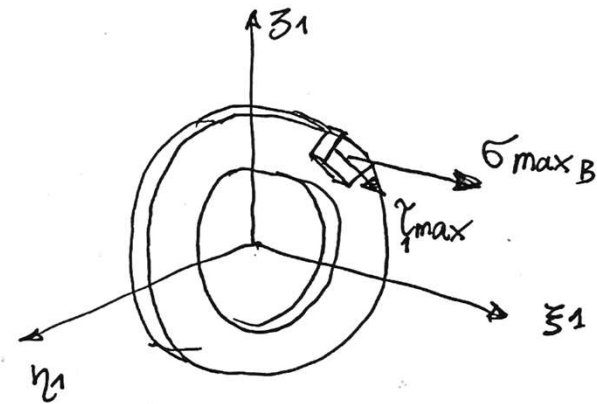
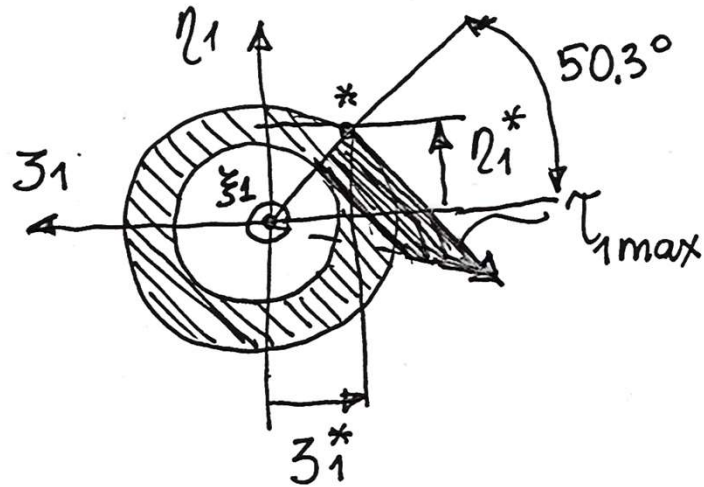
POINT OF THE HIGHEST STRESS :

ELEMENT  $\boxed{1}$  ,  $\xi_1 = 0$

NORMAL STRESS DUE TO BENDING :

$$\sigma_{\text{MAX B}} = \sigma_{\xi_1}^{\text{I}} + \sigma_{\xi_1}^{\text{II}} =$$

$$= - \frac{M_{z_1}(0) \cdot \eta_1^*}{J_{z_1}} - \frac{M_{\eta_1}(0) \cdot z_1^*}{J_{\eta_1}} = 161.9 \text{ MPa}$$



SHEAR STRESS

$$\tau_{1max} = \frac{E d_0 \alpha z}{4(1+\nu) L_1} = -38.86 \text{ MPa}$$

MAXIMUM EQUIVALENT STRESS :

$$\sigma_{EQU} = \sqrt{\sigma_{MAX B}^2 + 3 \tau_{1max}^2} = 175.34 \text{ MPa}$$

# Von Mises stress (SEQV) from ANSYS

